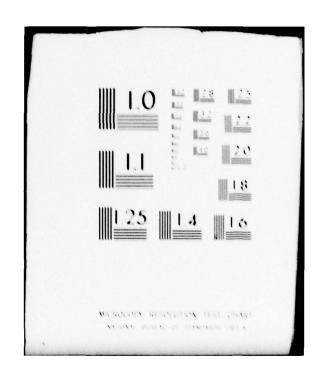
ARMY MISSILE COMMAND REDSTONE ARSENAL AL TECHNOLOGY LAB F/G 19/4 A COUPLED INTERIOR BALLISTICS FINITE ELEMENT COMBUSTION INSTABI--ETC(U) JUN 79 R M HACKETT DRDMI-T-79-68 UNCLASSIFIED NL OF AD 75110

END DATE FILMED DDC

AD-A075 110



**TECHNICAL REPORT T-79-68** 

A COUPLED INTERIOR BALLISTICS FINITE ELEMENT COMBUSTION INSTABILITY ANALYSIS PROCEDURE — PART II

Robert M. Hackett Technology Laboratory

26 June 1979



U.S. ARMY MISSILE COMMAND
Redetone Areenal, Alabama 35809

Distribution unlimited; approved for public release.

SMI FORM 1021, 1 JUL 79 PREVIOUS EDITION IS OBSOLETE

#### DISPOSITION INSTRUCTIONS

DESTROY THIS REPORT WHEN IT IS NO LONGER NEEDED. DO NOT RETURN IT TO THE ORIGINATOR.

#### DISCLAIMER

THE FINDINGS IN THIS REPORT ARE NOT TO BE CONSTRUED AS AN OFFICIAL DEPARTMENT OF THE ARMY POSITION UNLESS SO DESIGNATED BY OTHER AUTHORIZED DOCUMENTS.

#### TRADE NAMES

USE OF TRADE NAMES OR MANUFACTURERS IN THIS REPORT DOES NOT CONSTITUTE AN OFFICIAL ENDORSEMENT OR APPROVAL OF THE USE OF SUCH COMMERCIAL HARDWARE OR SOFTWARE.

Unclassified

REPORT DOCUMENTATION	PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM					
1. REPORT NUMBER	2. GOVT ACCESSION NO.	S. RECIPIENT'S CATALOG NUMBER					
т-79-68							
4. TITLE (and Subtitle)		S. TYPE OF REPORT & PERIOD COVERED					
A Coupled Interior Ballistics Fin Combustion Instability Analysis P	rocedure	Technical Report					
Part II	6. PERFORMING ORG. REPORT NUMBER						
7. AUTHOR(4)	6. CONTRACT OR GRANT NUMBER(e)						
Robert M. Hackett							
9. PERFORMING ORGANIZATION NAME AND ADDRESS		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS					
Commander US Army Missile Research and Deve ATTN: DRSMI-TK (R&D) Redstone Arsenal, Alabama 35809	lopment Command						
11. CONTROLLING OFFICE NAME AND ADDRESS		12. REPORT DATE					
Commander US Army Missile Research and Deve	lopment Command	26 June 1979					
ATTN: DRSMI-TI (R&D) Redstone Arsenal, Alabama 35809		40					
TA. MONITORING AGENCY NAME & ADDRESS(I differen	nt from Controlling Office)	18. SECURITY CLASS. (of this report)					
		Unclassified					
		154. DECLASSIFICATION/DOWNGRADING					
Distribution unlimited; approved	for public releas	e.					
17. DISTRIBUTION STATEMENT (of the abstract entered	in Block 20, if different fra	n Report)					
18. SUPPLEMENTARY NOTES							
19. KEY WORDS (Continue on reverse side if necessary a	nd Identify by block number)						
A computer code is formulate	d which provides	the solid propellent erain					
designer with the capability of p while, with a minimum amount of a	erforming an inte	rior ballistics analysis					
a combustion instability predicti	on analysis at an	y designated point in time					
during the entire period of performance. The program, in effect, couples the output of an existing solid propellant grain design evaluation code, which predicts the acoustic chamber geometry during surface regression, with the							
input to the existing three-dimen	sional finite ele	ment combustion instability					

ECURITY CLASSIFICATION OF THIS PAGE(Then D

20. prediction code, FLAP3. The three-dimensional finite element mesh and boundary conditions are generated from the grain surface regression data for the progressive burn times. The entire finite element mesh and boundary condition seneration by FLESH3, the commanion to FLAP3, is an

boundary condition generation by FLESH3, the companion to FLAP3, is executed with the input of seven geometry parameters, which are obtained from the ballistics code output, and the corresponding burn depth parameter. The use of the developed program is demonstrated for three different values of burn depth for the case of a typical star design.

Unclassified
SECURITY CLASSIFICATION OF THIS PAGE(When Date Enters

#### **ACKNOWLEDGMENT**

The assistance of Robert Radke in developing the computer code BRNMSH and in running the numerous test cases is gratefully acknowledged.

Appreciation is expressed to the MIRADCOM Propulsion Directorate, Redstone Arsenal, the Army Research Office - Durham, and Battelle Laboratories - Durham Office for financial support of this project.

## CONTENTS

Section	on Page
1.	Introduction 3
2.	Three-Dimensional Finite Element Mesh Generation
3.	Interior Ballistics Code 5
4.	Development of the BRNMSH Code 5
5.	Conclusions
	Appendix A-Listing of the Computer Program BRNMSH
	Appendix B-Example Three-Dimensional Finite Element
	Mesh Generation Input for Star Design27

## **ILLUSTRATIONS**

Figur	Pag	е
1.	Cross-Section of Repeating Segment of Star Design Showing	
	Design Parameters N, Re, W, Ya, a, L, and	5
2.	Cross-Section of Repeating Segment of Star Design Showing	
	Design Parameters N, R <sub>c</sub> , W, Y <sub>a</sub> , $\alpha$ and L; $\phi = 0$	6
3.	Cross-Section of Repeating Segment of Star Design Showing	
	the Four Progressive Burning Zones	7
4.	Cross-Section of Repeating Segment of Star Design Showing	
	the Four Progressive Burning Zones; $\phi = 0$	7
5.	Finite Element Mesh for Repeating Segment of Star Design	
	with Zone I Burning; $\phi = 0$	8
6.	Finite Element Mesh for Repeating Segment of Star Design	
	with Zone 2 Burning; $\phi = 0$	8
7.	Finite Element Mesh for Repeating Segment of Star Design	
	with Zone 3 Burning; $\phi = 0$	9

#### 1. INTRODUCTION

It has been recognized for some time that one of the major shortcomings of current combustion instability prediction technology is that it is computationally removed from conventional interior ballistic analysis. Traditionally, interior ballistics data have been extracted from an analysis and used to formulate, by hand, the problem to be run with one of the existing combustion instability prediction finite element codes. In an effort to initiate a remedy for this inefficient operation and thereby develop the basis for a more elaborate and meaningful design procedure for solid propellant rocket motors, the computer code GRNMSH (Grain Mesh) was developed. This code provided an automated procedure for generating the input to a finite element mesh generator from the output of an interior ballistics code. The code GRNMSH was designed to couple the output of the 564 Interior Ballistics Computer Program,2 developed by Aerojet-General Corporation, with the computer code FLESH3 (Fluid Mesh Generation, 3 Dimensions) which was developed as a companion finite element mesh generation program to the combustion instability prediction code FLAP3 (Fluid Analysis Program, 3 Dimensions). The development of GRNMSH was considered to be of a "proof of concept" nature and stopped short of providing the complete coupling of the two codes for the entire period of performance, i.e., for the duration of propellant burning, The GRNMSH code generates the input to FLESH3 only at the instant of ignition. This report is concerned with the extension of the GRNMSH code to enable the same automatic transfer of data between the two codes at any instant in time, from propellant ignition to burnout. The extension of GRNMSH has resulted in the development of the code BRNMSH (Burn Mesh) which generates the input to FLESH3 and thus initiates the generation of a coupled fluid-solid three-dimensional finite element mesh for any specified depth of burn.

The development has concentrated on the star design, which is depicted in *Figures 1* and 2. Similar designs can easily be handled in the same manner. A brief discussion of the previously developed codes, along with a description of BRNMSH follows.

<sup>1.</sup> R. M. Hackett, A Coupled Interior Ballistics—Finite Element Combustion Instability Analysis Procedure, US Army Missile Research and Development Command, Redstone Arsenal, Alabama, Technical Report T-78-72, July 1978.

A. E. Whetstone, T. R. Threewit, and J. S. Billheimer, Basic Grain Design and the 564 Interior Ballistics Computer Program, STM-143, Interior Ballistics Department, Applied Mechanics Division, Aerojet-General Corporation, Sacramento, California, June 1961.

<sup>3.</sup> R. M. Hackett and R. S. Juruf, A Three-Dimensional Finite Element Code for Combustion Instability Prediction. Proceedings of the 13th JANNAF Combustion Meeting, Naval Postgraduate School, Monterey, California, CPIA Publication 281, December 1976.

#### 2. THREE-DIMENSIONAL FINITE ELEMENT MESH GENERATION

The FLAP3 code performs a linear acousto-modal analysis of the irrotational motions of an inviscid, compressible fluid coupled to the motion of a nearly incompressible, linearly viscoelastic solid, and a linear potential flow analysis of the irrotational motions of an inviscid, incompressible fluid, and then determines the effect of the flow field and of combustion on the calculated acoustic oscillations. This combustion instability analysis is performed at different points in time, beginning at ignition, or time zero. The output of FLAP3 is modal frequency and an evaluation of the pressure growth/decay coefficient for each mode of vibration analyzed. A positive net value of the coefficient indicates a growth of pressure oscillations and, therefore, instability while a negative value of the coefficient is an indication of decaying oscillations, or stability.

The FLAP3 analysis utilizes the finite element method and models the acoustic cavity and propellant grain as an assemblage of three-dimensional quasi-hexahedral (each hexahedron is a combination of five tetrahedra) elements connected at the corners. In order to be realistically applicable, any three-dimensional finite element code requires a companion mesh generator; a coupled fluid-solid analysis code requires a mesh generator of more than the customary complexity. The code FLESH3 was designed to be used in conjunction with FLAP3. A detailed description of the use of FLESH3 is found in Hackett's User's Manual for FLAP3.4 The code FLESH3 generates the numbered nodal points and their coordinates and identifies each node as to whether it lies in the acoustic cavity, on the cavity-grain interface, or in the grain; generates the quasi-hexahedral elements, designated by the eight numbered nodal points defining each element, and identifies each element as to whether it is a cavity element, a cavity element adjacent to the cavity-grain interface, or a solid propellant element; and generates the cavity-grain interface element surfaces (burning surfaces) and identifies each as to the direction of the surface normal. This information, along with a few additional input parameters relative to gas and propellant properties and type of acoustic mode(s) to be analyzed, is necessary and sufficient to activate a combustion instability analysis by FLAP3, given adequate computational facilities.

The input to FLESH3 is in the form of a set of global curves, defining cavity and grain boundaries; a designation of a sequence of points, as to whether they are cavity, interface or grain points; a designation of blocks of elements to be generated, by the nodal indices of the blocks; and a designation of the number and location of longitudinal cross-sections, establishing the number of layers of elements.

<sup>4.</sup> R. M. Hackett, *User's Manual for FLAP3*, US Army Missile Research and Development Command, Redstone Arsenal, Alabama, Technical Report TK-77-4, July 1977.

The objective is that of minimizing the amount of time required to prepare the input for FLAP3 and the likelihood of error in the preparation of that data. Through the use of BRNMSH, which will be subsequently described, all of the necessary geometry and identification input data for FLAP3 can be generated with the input of seven geometry parameters, for the star design, plus one additional combustion parameter.

#### 3. INTERIOR BALLISTICS CODE

The interior ballistics analysis code used in the formulation of BRNMSH was developed by Aerojet-General Corporation.<sup>2</sup> The BRNMSH geometrical input parameters are, therefore, those presented in that code, but similar parameters would be obtained from a consideration of a comparable interior ballistics code. The description of BRNMSH will be related to the Aerojet code, but the procedure followed in the development is general.

#### 4. DEVELOPMENT OF THE BRNMSH CODE

A listing of the formulated computer code BRNMSH is found in Appendix A. The code is based upon the star design found in the report by Whetstone et al.<sup>2</sup> Due to the dihedral symmetry provision in FLAP3, only the smallest repeating segment need be analyzed. This segment, for the general star geometry, is shown in Figures 1 and 2; with the independent geometry parameters identified. The only difference between Figure 1 and Figure 2 is that the angle  $\phi$  is zero in Figure 2. This condition necessitated a slightly different internal provision in BRNMSH. Figures 3 and 4 define the different burning zones and the regression of the interface surface with, again, the difference in the two figures being the value of the angle  $\phi$ . Figure 5 shows a typical finite element mesh for burning in Zone 1. The cross-hatched portion of the segment represents the grain elements. The mesh is seen to be similar to that which would be generated for the case of zero burning. Figure 6 shows a typical mesh for burning in Zone 2. It can be noted that the generated mesh for Zone 2 burning differs distinctly from that for Zone 1 burning. Figure 7 shows a typical mesh for burning in Zone 3. As the burn depth increases it can be seen that the cavity geometry becomes more simply defined and this fact is thus reflected in the generated meshes of the changing cavity-grain cross-section.

A working knowledge of FLESH3<sup>4</sup> is necessary in order to understand the intricacies of BRNMSH, and to be able to modify and add to it, but not in order to use it. An understanding of the use of BRNMSH can be enhanced through the consideration of an example. *Table 1* lists the twelve cases of the star design upon which the formulation of BRNMSH is based. Design No. D-6c is chosen for the example. Three different burn depths are selected. They are: 0.05 in. (burning in Zone 1), 1.50 in. (burning in Zone 2), and 3.00 in. (burning in Zone

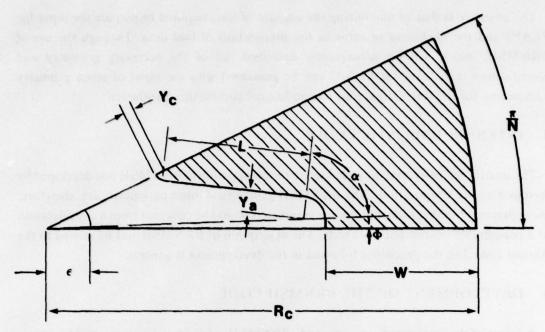


Figure 1. Cross-section of repeating segment of star design showing design parameters N, R<sub>C</sub>, W, Y<sub>a</sub>,  $\alpha$ , L and  $\Phi$ .

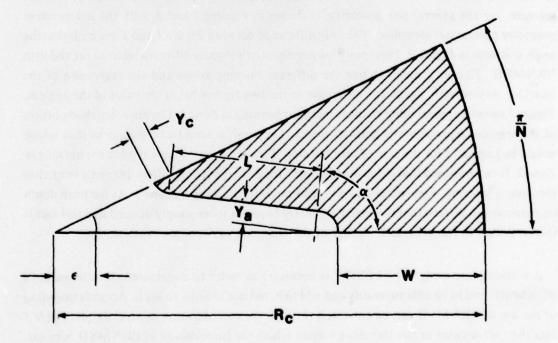


Figure 2. Cross-section of repeating segment of star design showing design parameters N, R<sub>C</sub>, W, Y<sub>a</sub>,  $\alpha$  and L;  $\Phi$ = 0.

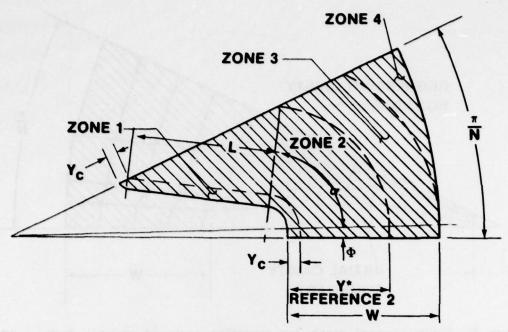


Figure 3. Cross-section of repeating segment of star design showing the four progressive burning zones.

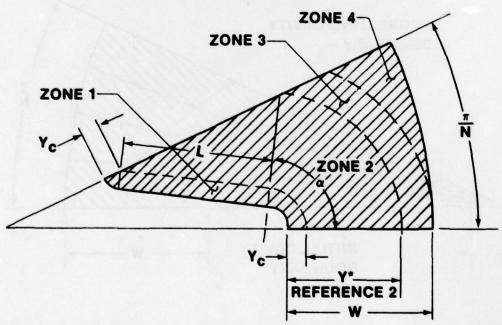


Figure 4. Cross-section of repeating segment of star design showing the four progressive burning zones;  $\Phi$  = 0.

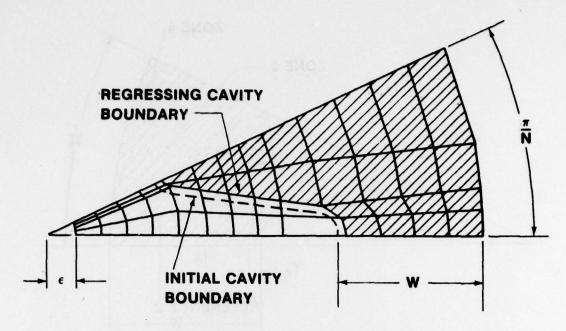


Figure 5. Finite element mesh for repeating segment of star design with Zone 1 Burning;  $\Phi$  = 0.

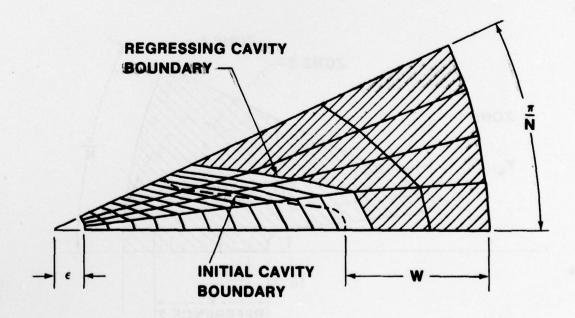


Figure 6. Finite element mesh for repeating segment of star design with Zone 2 Burning;  $\Phi$  = 0.

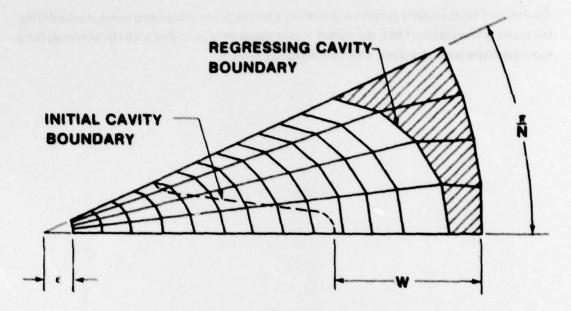


Figure 7. Finite element mesh for repeating segment of star design with Zone 3 Burning;  $\Phi=0$ .

3). Each case requires the input of two data cards to BRNMSH. The first eard contains the seven parameters: N, R, W,  $\phi$ , Y,  $\alpha$ , and L (in that order). The second card contains the burn distance, BURN. The output for the three separate cases, which is the input to FLESH3, is shown in Appendix B.

It can be noted that the length of the grain, and provision for five equidistant sections are conditions which are internally set in the program, but could, alternatively, be input as an additional set of parameters. It can also be noted that the value of  $\epsilon$ , the necessary offset radius of the nodes closest to the axis origin, is internally set at 0.1 in.

There are other modifications which could be made to BRNMSH in order to increase its flexibility, for instance that of developing the capability of generating a mesh for a tapered longitudinal profile. Possible modifications such as this are obvious, but other more subtle changes could also be made which would increase the generality of the code.

#### 5. CONCLUSIONS

A computer code has been developed which couples an interior ballistics analysis code with a combustion instability analysis code and facilitates the efficient formulation of the three-

dimensional finite element model for instability analysis at any designated point in time during the entire performance. Thus, the rocket motor designer is provided with the potential for a more inclusive and, therefore, superior design approach.

In an alrest travelling to said not not your base many out to a running the content of the city

TABLE 1. STAR DESIGN PARAMETERS-SYMMETRY 7 (N = 7)

STAR (Design No.)	R <sub>c</sub>	w in.	α deg	Ф deg	Ya In.	Y <sub>C</sub>	L in.
D-5a	10	3.5	80	0.0	0.2	0.075	4.3
D-5b	10	3.5	85	0.0	0.2	0.068	4.9
D-5c	10	3.5	75	0.0	0.2	0.136	3.8
D-6a	10	3.5	85	0.0	0.5	0.092	4.1
D-6b	10	3.5	80	0.0	0.5	0.118	3.6
D-6c	10	3.5	75	0.0	0.5	0.095	3.3
D-7a	10	3.5	85	2.5	0.2	0.043	4.8
D-7b	10	3.5	80	2.5	0.2	0.084	4.1
D-7c	10	3.5	75	2.5	0.2	0.047	3.7
D-8a	10	3.5	85	2.5	0.5	0.092	3.9
D-8b	10	3.5	80	2.5	0.5	0.035	3.5
D-8c	10	3.5	75	2.5	0.5	0.069	3.1

# APPENDIX A LISTING OF THE COMPUTER PROGRAM BRNMSH

```
PROGRAM PRNMSH
     ACINPUT. DUTPUT. PUNCH. TAPES=INPUT. TAPES=OUTPUT. TAPE7=PUNCH)
      PROGRAM BRYMSH
     A (INPUT. DUTPUT. PUNCH. TAPES=INPUT. TAPE6=OUTPUT. TAPE7=PUNCH)
COMMENT ...
COMMENT...THIS ROUTINE DEVELOPES THE INPUT FOR FLESHS
COMMENT ... STATUS ---- OPERATIONAL ----- 18MAY79
COMMENT ...
      DIMENSION X(90). Y(90). AA(90). PB(90). THI(90). THF(90)
      DIMENSION NN(90) . AP(90)
      DIMENSION IEND(3)
      DATA LUI/5/. LUO/6/. LUA/7/
      DATA IEND(1)/1HE/+ IEND(2)/1HN/+ IEND(3)/1HD/
DATA PI/3-14159265/+ PID/1P0-0/+ DEGRAD/1-74532925E-02/
      DATA ZERO/0.0/
      DATA IZERO/0/ NC/50/
      DATA ITITLE/10HGRAIN MESH/
 6111 FORMAT(A10)
      WRITE (LUA. 6111) ITITLE
 5211 FORMAT(7510.2)
      READ
            (LUI-5211) XN+ RC+ W+ PHID+ YA+ ALPHAD+ XL
5311 FORMAT (F10.2)
      READ (LUI.5311) BURN
      PHI=PHIC+DEGRAD
      ALPHA=ALPHAD+DEGRAD
      THETAD=PID/XN
      THETA=THETAD+DEGRAD
      IMAX=14
      JMAX=5
      NNC=39
      IBC=3
      SCALE=1.0
      NLAY=5
      Z=ZERO
      DZ=2.0
      C.B=XAPS
      DO 111 I=1.NC
      X(I)=ZERO
      Y(I)=ZERO
      NN(I)=IZERC
      NP(I)=IZERO
      AA(I)=ZERC
      BB(I)=ZERO
      THI(I)=ZERO
      THF (1)=ZERO
  111 CONTINUE
      EPSLON=0.1
      RCWYA=RC-W-YA
      PINPHI=THETA-PHI
      SFPHI=SIN(PINPHI)
      CFPHI=COS(PINPHI)
      DELTA=PI+0.5+THETA-PHI-ALPHA
      SFDEL=SIN(DELTA)
      CFDEL=COS(DELTA)
      SPHI=SIN(PHI)
      CPHI=COS(PHI)
      SFAP=SIN(PHI+ALPHA)
      CFAP=COS (PHI+ALPHA)
      BETA=PI+0.5-PHI-ALPHA
      SBETA=SIN(PETA)
      CBETA=COS(BETA)
      APT=ALPHA+0.5+PHI+0.5+THETA
      SAPT=SIN(APT)
      CAPT=COS(APT)
```

```
YC=(RCWYA+SFPHI-YA+CFDEL-XL+SFDEL)/SINCALPHA+PHI-THETA)
     YCB=YC-BURN
     YSTAR=RCWYA+SFPHI/SIN(ALPHA+PHI-THETA)-YA
     Y9=YA+BURN
     B2=9URN-YC
     93=BURN-YSTAR
     H=(YSTAR-YC-B2)+SIN(ALPHA-THETA)/SFDEL
    H1=(YSTAR-YC-32)+SIN(ALPHA+PHI-THETA)/SFDEL
     IF (PHI.LT.THETA.AND.BURN.LT.YC) IBC=7
6211 FORMAT(415. F10.2.15)
    WRITE (LUA.6211) IMAX. JMAX. NNC. IRC. SCALE. NLAY
 201 CONTINUE
     I=1
     NN(I)=1
     NP(I)=THETAD
    AA(I)=EPSLON
    BB(I)=EPSLON
     THF(I)=THETAD
202 CONTINUE
    1=3
     X(I)=RC
     NN(I)=2
203 CONTINUE
     1=4
     NN(1)=3
     NP(I)=THETAD
     AA(I)=RC
    BB(I)=RC
     THF(I)=THETAD
204 CONTINUE
    1=6
    X(I)=RC
     Y(I)=RC+TAN(THETA)
    NN(I)=4
205 CONTINUE
    I=7
    NN(I)=5
    NP(I)=THETAD
     AA(I)=RC-W
    SHII)=AAII)
    THF(I)=THETAD
206 CONTINUE
    1=6
    CC=(RCWYA+SPHI+Y8+SFAP+XL+SBETA)-(RCWYA+SPHI+YB+SFAP)
    DD=(RCWYA+CPHI+Y8+CFAP)-(RCWYA+CPHI+Y8+CFAP-XL+CBETA)
    SLOPE1=CC/DD
    X(I)=RCWYA+CPHI+YB+CFAP-XL+CBETA-0.1
    Y(I)=RCWYA+SPHI+YB+SFAP+XL+SPETA+0.1+SLOPE1
    1=9
    X(I)=RCWYA+CPHI+YB+CFAP+0.1
    Y(I)=RCWYA+SPHI+YB+SFAP-0.1+SLOPE1
    NN(I)=6
207 CONTINUE
    I=10
    X(I)=RCWYA+CPHI
     Y(I)=RCWYA+SPHI
     NN(T)=7
     VP(I)=ALPHAD
    AA(I)=YE
    38(1)=Y5
     THI(I)=PHIO
    THF(I)=PHID+ALPHAD
208 CONTINUE
    I=12
    GGG=RCWYA+SPHI+YB+SFAP+XL+SAETA
    HHH=RCWYA+CPHI+YB+CFAP-XL+CRETA
```

```
SLOPE2=GGG/HHH
    X(1)=RCWYA-CPHI+Y8-CFAP-XL-CBETA+0.1
    Y(1)=RCWYA+SPHI+YP+SFAP+XL+SPETA+0.1+SL OPE2
    NN(I)=8
209 CONTINUE
    1=13
    X (I) = RCHYA + CPHI + YP + CFAP - XL + CFETA + YCB + SBETA
    Y(I)=RCWYA.SPHI.YB.SFAP.XL.SSETA.YCB.CBCTA
    NN(1)=9
    NP(I) = ALPHA 2 + PHID - THE TAD
    AA(1)=YCP
    88(1)=YC9
    THICI)=PID+THETAD-1.0
    THF (1) =PID+PHID+ALPHAD+1.0
21C CONTINUE
    1=15
    X(I)=RC
    Y(1)=RC+TAN(PHI)
    NN(1)=10
211 CONTINUE
    1=16
    NN(1)=11
    CAT 3HT = (I) 9N
    AA(I)=((RCWYA+CFPHI+YB+SFDEL)++2+(RCWYA+SFPHI-YB+CFDEL)++2)++0.5
    BB(I)=AA(I)
    THF(I)=THETAD
212 CONTINUE
    1=19
    X(I)=RC
    Y(1)=RC+TAN(THETA+0.5)
    NN(I)=12
213 CONTINUE
    1=19
    NN(I)=13
    NP(I)=THETAD
    CCC=(RCWYA+CFPHI+YB+SFDEL-YL+CFDEL)++2
    DOD=(RCWYA+SFPHI-YB+CFDEL-XL+SFDEL)++2
    AA(I)=(CCC+DDD)++0.5
    BB(I)=AA(I)
    THF(1)=THETAD
214 CONTINUE
    1=20
    EE=AA(19) +S IN(THETA+0.5)-AA(16) +SPHI
    FF=AA(16) + CPHI - AA(19) + COS(THETA+0.5)
    SLOPE3=EE/FF
    X(1)=AA(19) +COS(THETA+0.5)-0.1
    Y(1)=AA(19) .SIN(THETA.0.5)+0.1.SLOPE3
    1=21
    X(1)=AA(16)+CPHI+0-1
    Y(1)=AA(16)+SPHI-0.1+SLOPES
    NN(I)=14
215 CONTINUE
    1=22
    NN(I)=15
    NP(I)=THETAD
    AA(I)=EPSLON+(AA(19)-AA(1))+0.75
    BB(I)=AA(I)
    THF(1)=THETAD
216 CONTINUE
    1=23
    GG=AA(16)+SPHI-AA(7)+SIN(PHI+0.5)
    HH=AA(7) +COS(PHI+0.5)-AA(16)+CPHI
    SLOPE4=GG/HH
    X(1)=AA(16)+CPHI-0.1
    Y(1)=AA(16)+SPHI+0.1+SLOPE4
    1=24
```

```
X(1)=AA(7) . COS(PHI . 0.5) + 0.1
    Y(1)=AA(7) .SIN(PHI-0.5)-0.1.SLOPE4
    NNTI)=16
217 CONTINUE
    1=25
    NN(I)=17
    NP(I)=THETAD
    AA(I)=AA(19)+(AA(16)-AA(19))+0.75
    BB(I)=AA(I)
    THE CIDETHETAD
218 CONTINUE
    1=27
    X(I)=RC
    Y(I)=RC+TAN(PHI+0.5)
    NN(T)=18
219 CONTINUE
    1=28
    VV=RC+(YB+SIN(ALPHA)/(RCHYA+YB+COS(ALPHA)))-RCHYA+SPHI+YB+SFAP
    WWERE-RCJYA-CPHI-YP-CFAP
    SLOPES=VV/WW
    X(I)=RCWYA+CPHI*YB+CFA>-0.1
    YII)=RCWYA+SPHI+YP+SFAP-0.1+SLOPE5
    1=23
    X(I)=RC
    Y(1)=RC+(Y9+SIN(ALPHA)/(RCWYA+Y9+COS(ALPHA)))
    NN(1)=20
221 CONTINUE
222 CONTINUE
    1=31
    X(I)=RC
    Y(I)=(RC+(YB+SIN(ALPHA)/(RC+YA+YE+COS(ALPHA))))+0.5
    NN(1) =22
223 CONTINUT
224 CONTINUE
    1=32
    UUU=Y(31) ** 2/(RC ** 2)
    WW=(X(10)+((Y9++2)+(1.0+UUU)-(X(10)++2)+UUU)++0.5)/(1.0+UUU)
    222=660-7131)/AC
777=AA(19)-SIN(THETA-0.5)-222
    XXX=WW-AA(19)+COS(THETA+0.5)
    SLOPE6=YYY/XXX
    X(1)=AA(19)+COS(THETA+0.5)-C.1
    Y(1)=AA(19) +SIN(THETA+0.5)+0.1+SLOPE6
    1=33
    X(1)=WWW+0.1
    Y(1)=222-0.1.SLOPES
    NN(1)=24
225 CONTINUE
225 CONTINUE
    1=35
    SLOPE 7= (Y(13)-YC8+SAPT)/(X(13)-YC8+CAPT)
    X(1)=X(13)-YCB+CAPT+0+1
    Y(1)=Y(13)-YC8+SAPT+0.1+SLOPF7
    NN(1)=26
227 CONTINUE
22º CONTINUE
    1=36
    X(1)=X(13)-YC8+CAPT-0.1
    Y(1)=Y(13)-YC8+SAPT-0.1+TAN(THETA+0.25)
    1=37
    X(1)=AA(16)
    Y(1)=Y(13)-YC8+SAPT+(AA(16)-X(13)+YC3+CAPT)+TAN(THETA+0.25)
    NN(1)=28
229 CONTINUE
230 CONTINUE
```

```
1=33
    AD=X(13)-YC++CAPT
    AE=Y(13)-YCH+SAPT
    AF=AA(15)
    AS=TANCTHETA-0.25)
    AX=46 ... 2.0+1.0
    AY=2.0. (AD+AE+AG)
    42=40 . 2 . 0 . 4E . . 2 . 0 - 4F . . 2 . 0
    (XA+0-5)\(6.0+(5A+XA+0-4-0-5)\(2.0+AX)
    BW=AU+AG
    CW=AA(7)+SIN(THETA+0.5)-(Y(13)-YC4+SAPT+BW)
    DW=AA(7) . COS(THETA . 9. 5) - (X(13) - YC9 . CAPT . AH)
    SLOPE 8=CW/DW
    X(1)=X(13)-YCR+CAPT+AW-0.1
    Y(1)=Y(13)-YC8+SAPT+BW-0.1+SLOPF9
    1=19
    X(1)=AA(7).COS(THETA.0.5).0.1
    Y(1)=AA(7).SIN(THETA.0.5).0.1.SLOPER
    NN(1)=30
231 CONTINUE
232 CONTINUE
    1=49
    X(1)=RC+0.25
    Y(1)=RC+0.25+TAN(THETA+0.5)
    1=41
    X(I)=RC
    Y(I)=RC+TAN(THETA+0.5)
    NN(1)=32
233 CONTINUE
234 CONTINUS
    1=43
    SP=(I)X
    Y(1)=RC+TAN(THETA+0.25)
    NN(1)=34
235 CONTINUE
236 CONTINUE
    1=45
    X(1)=RC
    Y(I)=RC+TAN(THETA+0.50)
    VN(I)=16
237 CONTINUE
238 CONTINUE
    1=47
    X(I)=QC
    Y(I)=RC+TAV(THETA+0.75)
    NN(1)=18
239 CONTINUE
240 CONTINUE
    1=48
    AB=AA(16)+SIN(THETA+0.5)-(Y(13)-YC3+SAPT)
    AC=AA(14) + COS(THETA+0.5) - ( V(13) - YCE + CAPT)
    SLOPE 9= AB/AC
    X(1)=X(13)-YC9+CAPT-0.1
    Y(1)=Y(13)-YC5+SAPT-0.1+SLCFE9
    1=49
    X(1)=AA(16) . COS(THETA.0.5)+0.1
    Y(1)=AA(15)+SIN(THETA+0.5)+0.1+SLOPE9
    NN(1)=40
241 CONTINUE
242 CONTINUE
    1=50
    X(1)=RCWYA+(YA+YC+82)
    1=51
    X(1)=RCWYA+(YA+YC+32)+COS(ALPHA/3.0)
    Y(1)=(YA+YC+92)+SIN(ALPHA/3.0)
    1:52
```

```
X(1)=RCWYA+(YA+YC+321+COS(ALPHA+2.0/3.0)
    Y(1)=(YA+YC+32)+SIN(ALPHA+2.0/3.0)
    1=53
    X(I)=RCUYA+(YA+YC+B2)+COS(ALPHA)
    Y(1)=(YA+YC+82)+SIY(ALPHA)
    1=54
    X(1)=X(53)-H-COS(PI-0.5-ALFHA)/2.0
    Y(1)=Y(53)+4-SIN(PI+0.5-ALPHA)/2.0
    X(1)=X(53)-H.C3S(PI.C.5-ALPHA)
    Y([)=Y(53)+H+SIN(PI+0.5-ALP-A)
    NN(1)=42
243 CONTINUS
    1=06
    X ( I ) = R C & Y A
    NN(I)=45
    NP(I) = ALPHAD
    AA(1)=YA+YC+83
    BR(I)=AA(I)
    THE (1) = ALPHAD
244 CONTINUE
    1=57
    x(1)=x(50)/1.125
    1=58
    X(1)=X(51)/1.125
    Y(1)=Y(51)/1.125
    1=50
    X(1)=X(52)/1.125
    Y(1)=Y(52)/1-125
    1=60
    X(1)=X(53)/1.125
    Y(1)=Y(53)/1.125
    1=61
    X(1)=X(54)/1.125
    Y(1)=Y(54)/1-125
    1=62
    X(1)=X(55)/1.125
    Y(1)=Y(55)/1.125
    NN(1)=40
245 CONTINUE
    1=63
    X(I)= CUYA
    NN(1)=45
    NP(1)=ALPHAD
    AA(1)=AA(56)/1-125
    BB(I)=AA(I)
    THE ( 1) = ALPHAD
246 CONTINUE
    1=64
    X(I)=RCdYA+(YA+YC+R2)
    1:65
    X(1)=(RC-W+YC+52)+CPHI
    Y(1)=(RC-W+YC+82)+SPHI
    X(1)=QCUYA+CPHI+(YA+YC+B2)+COS(PHI+ALPHA/3.0)
    Y(I)=RCUYA+SPHI+(YA+YC+R2)+SIN(PHI+ALPHA/3.0)
    X(1)=RC+YA+CPHI+(Y4+YC+E2)+COS(PHI+ALPHA+2.0/3.0)
    Y(I)=RCKYA+SPHI+(Y4+YC+32)+SIN(PHI+ALPH4+2.0/3.0)
    X(I)=RCWYA+CPHI+(YA+YC+R2)+CFAC
    Y(1)=9C4YA+SPHI+(Y4+YC+92)+SFAP
    1=69
    0.5/AT3F3-14-(88)X=(1)X
    C.S\ATBEZ+1H+(88)Y=(1)Y
    1=70
```

```
X(1)=X(68)-H1-CFETA
     Y(1)=Y(68)+H1-S3ETA
     N4(1)=44
 247 CONTINUE
     I=71
     NN(I)=47
     NP(I)=PHIC
     AA(1)=#0-4+YC+23
     SE(I)=AA(I)
     THF(I)=PHIO
 248 CONTINUE
     1=72
     x(1)=X(64)/1.125
     1=75
     x(1)=x(65)/1.125
     Y(1)=Y(65)/1.125
     I=74
     x(1)=x(66)/1.125
     Y(1)=Y(66)/1.125
     1=75
     x(1)=x(67)/1.125
     Y(1)=Y(67)/1.125
     1=75
     X(1)=X(68)/1.125
     Y(1)=Y(68)/1.125
     1=77
     x(I)=x(69)/1.125
     Y(1)=Y(69)/1.125
     1=78
     X(1)=X(70)/1.125
     Y(1)=Y(70)/1.125
     NN(I)=48
 249 CONTINUE
     1=79
     X(I)=RCWYA+CPHI
     Y(I)=RCWYA+SPHI
     NN(1)=49
     NP(I)=ALPHAD
     AA(I)=YA+YC+B3
     88(I)=AA(I)
     THF (I) = ALPHAD
 250 CONTINUE
 251 CONTINUS
     1=80
     NN(I)=51
     NP(I)=PHIO
     AA(I)=AA(71)/1.125
     BB(I)=AA(I)
     THF(I)=PHID
252 CONTINUE
 253 CONTINUE
     1=81
     X(I)=RCWYA+CPHI
     Y(I)=RCWYA+SPHI
     NN(1)=53
     NP(I)=ALPHAD
     AA(I)=AA(79)/1.125
     BP(I)=AA(I)
     THF (I) = ALPHAD
 299 CONTINUE
     00 331 I=1.91
     WRITE (LUA-6331) X(I)-Y(I)-NN(I)-NP(I)-AA(I)-BB(I)-THI(I)-THF(I)
331 CONTINUE
6331 FORMAT(2F10.4,215,4F10.4)
     141=-1
     II=ZERO
```

```
13=1
      IK=2
       IL=3
      IM=4
      IN=5
      19=6
      10=7
      IREF
      15=3
      J1=10
      JJ=11
      JK=12
      JL=15
      JM=14
      JN=15
      JP=16
      JQ=17
      JR=19
      JS=19
      KI=20
      KJ=21
      KK= 22
      KL=23
      KM=24
      KN=25
      KP=26
      KQ=27
      KR=28
      <5=29
      L1=30
      LJ=31
      LK=32
      LL=33
      LM= 14
      LN=35
      LP= 16
      LQ=37
      LR = 38
      LS= 19
      PI=40
      MJ=41
      MK=42
      ML = 43
      MM=44
      WN=45
      VP=46
      ¥9=47
      48=48
      MS=49
      NI =50
      NJ=51
      NK=52
      NL=53
      120VE=0
      IF(FURN.LT.YC) IZONE=1
      IF (PURN.GE. YC .AND. BURN.LT.YSTAR) IZONE=2
IF (-URM.GE.YSTAR .AND. BURN.LE.W) IZONE=3
IF (IZONE.NE.O) GO TO 4+0
 479 CONTINUE
WRITE (LUO.6439) SURN.YC.YSTAR./
6439 FORMAT (114 ZONE ERROR ./.4(1x.F10.4))
GO TO 999
 440 CONTINUE
      IF (PHI-LT-THETA) SO TO 442
6373 FORMAT (515)
      WRITE (LUA-6373) IJ-IJ-IJ-IS-IN
```

```
WRITE (LUA.6373) IX.JI.IJ.JI.IV
     WRITE (LUA.6373) IL.JJ.IJ.J.J.IV
441 CONTINUS
6443 FORMAT (1415)
     WRITE (LUA-6443) IJ.IJ.IJ.IS.IK.IK.JJ.LM.IJ.IJ.II.II.II.II
     WRITE (LUA-6443) IKOIJOIKOISOILOLMOJJOLPOIJOIJOIIOIIOIIOII
     WRITE (LUA.6443) IL. IJ. IL. IS. IM. LP. JJ. LR. IJ. IJ. II. II. II.
     WRITE (LUA.6443) IM.IJ.IM.IS.IN.LR.JJ.IM.IJ.IJ.II.II.II.II
     WRITE (LUA.5445) IV. IS. IJ. JI. IK. IN. L. M. JJ. IK. 11. 11. 11. 11
     WRITE (LUA.6443) IP.IS.IK.JI.IL.LM.IN.LP.JJ.IK.II.II.II.II
     WRITE (LUA-6443) IS-IS-IL-JI-IM-LP-IN-LR-JJ-IK-II-II-II-II
     WRITE (LUA.6443) IR. IS. IN. JI. IN. LR. IN. IM. JU. IK. II. II. II. II
     WRITE (LUR. 6443) ISOJI. IJOJMO IKO IKO ILOL MO INO ILO ITO ITO ITO ITO ITO
     WRITE (LUA-6443) JI-JI-IK-JM-IL-LM-IL-LP-IN-IL-II-II-II-II
     WRITE (LUA-6443) JJ.JI.IL.JM.IM.LP.IL.LR.IN.IL.II.II.II.II.II
     WRITE (LUA-6443) UK-JI-IM-JM-IN-LR-IL-IM-IN-IL-II-II-II-II
     50 TO 450
442 CONTINUS
     IF (IZONE.NE.1) GO TO 444
6332 FORMAT (515)
     WRITE (LUA.6332) IJ.IJ.IJ.IM.I"
     WRITE (LUA.6332) IJ.IN.IJ.IS.IK
     WRITE (LUA.6332) IK.IN.IL.IN.IN
     WRITE (LUA.6332) IK. IP.IL.IS.IL
     WRITE (LUA.6332) IK.JI.IJ.JI.IL
     WRITE (LUA.6332) IL.IP.IM.JM.IN
WRITE (LUA.6332) IL.JJ.IJ.JM.IL
 443 CONTINUE
     IF (PHI.EQ. 0.0) 30 17 445
6441 FORMAT(1415)
     WRITE (LUA-6441) IJ. IJ. IJ. IV. IK. IK. JL. JK. IJ. IJ. II. II. II.
     WRITE (LUA-6441) IK-IJ-IK-IN-IL-JK-JL-IR-IJ-IJ-II-II-II-II
     WRITE (LUA-6441) IL. IJ. IL. IM. IM. IR. JN. KP. IJ. IJ. II. II. II. II
     WRITE (LUA.6441) IM. IJ. IM. IM. IM. IN. KP. JN. IM. IJ. IJ. II. II. II. II
     WRITE (LUA-6441) IN-IN-IJ-IS-IK-IK-JJ-JM-JL-IJ-II-II-II-II
     WRITE (LUA.6441) IP.IM.IL.IN.IM.IR.IS.KP.JN.IK.II.II.II.II
     WRITE (LUA.6441) ID.IM.IM.IN.IN.KP.IS.IM.JN.IK.II.II.II.II.II
     WRITE (LUA-6441) IR.IN.IK.IS.IL.JM.JJ.IP.JL.IK.II.II.II.II
     WRITE (LUA,6441) IS.IS.IJ.JI.IK.IK.IN.JP.JJ.IK.II.II.II.II
     WRITE (LUA-6441) JIOISOIKOJIOILOJPOINOIGOJJOIKOIIOIIOIIOII
     WRITE (LUA,6441) JJ. IN. IL. IS. IM. IP. JJ. KR. IS. IL. II. II. II. II
     WRITE (LUA. 6441) JK. IN. IM. IS. IN. KR. JJ. IM. IS. IL. II. II. III. II
     WRITE (LUA-6441) JL-IS-IL-JI-IM-10-IN-LI-JJ-IL-II-II-II-II
     WRITE (LUA-6441) JM.IS.IM.JI.IN.LI.IN.IM.JJ.IL.II.II.II.II
     WRITE (LUA-6441) JN.JI.IJ.JM. IK.IK.IL.JR. IN.IL.II.II.II.II
     WRITE (LUA-6441) JP.JI.IK.JM.IL.JR.IL.JI.IN.IL.II.II.II.II
     WRITE (LUA-6441) JQ-JI-IL-JM-IM-JI-IL-LK-IN-IL-II-II-II-II
     WRITE (LUA-6441) JR.JI.IM.JM.IN.LK.IL.IM.IN.IL.II.II.II.II.II
     GO TO 450
445 CONTINUS
6442 FORMAT (1415)
     WRITE (LUA-6442) IJ.IJ.IJ.IN.IK.IK.JL.JK.IJ.IJ.II.II.II.II
     WRITE (LUA+6442) IK+IJ+IK+IN+IL+JK+JL+IR+IJ+IJ+II+II+II+II
     WRITE (LUA.6442) IL. IJ. IL. IM. IM. IR. JN. KP. IJ. IJ. II. II. II. II
     WRITE (LUA.6442) IM.IJ.IM.IM.IN.KP.JN.IM.IJ.IJ.II.II.II.II
     WRITE (LUA.6442) IN. IN. IJ. IS. IK. IK. JQ. KM. JL. IJ. II. II. II
     WRITE (LUA.6442) IP.IM.IL.IN.IM.IR.IS.KP.JN.IK.II.II.II.II
     WRITE (LUA-5442) IG.IM.IM.IN.IN.KP.IS.IM.JN.IK.II.II.II.II.
     WRITE (LUA-6442) IR.IN.IK.IS.IL.KM.JQ.IP.JL.IK.II.II.II.II
     WRITE (LUA+6442) IS+IS+IJ+JI+IK+IK+IQ+KH+JQ+IK+II+II+II+II+II
     WRITE (LUA.6442) JI.IS.IK.JI.IL.KM.IQ.IF.JQ.IK.II.II.II.II
     WRITE (LUA.6442) JJ. IN. IL. IS. I. T. P. JQ. MI. IS. IL. II. II. II. II
     WRITE (LUA.6442) JK.IN.IM.IS.IN.MI.JG.IM.IS.IL.II.II.II.II.II
     WRITE (LUA.6442) JL.IS.IL.JI.IM.IP.JJ.MI.JG.IL.II.II.II.II
```

```
WRITE (LUA-6442) JM-IS-IM-JI-IN-MI-JJ-IM-J7-IL-II-II-II-II
    WRITE (LUA.6442) JNOJIOIJOJMOIKOIKOILOKKOIZOILOIIOIIOII
    WRITE (LUA.6442) JP.JI.IK.JM.IL.KK.IL.KI.IQ.IL.II.II.II.II
    WRITE (LUA-6442) JR.JI.IM.JM.IN.LK.IL.IM.JJ.IL.II.II.II.II
    GO TO 450
 444 CONTINUE
    1F(120NE.NE.2) 50 TO 446
6334 FORMAT (515)
    WRITE (LUA-6334) IJ-IJ-IJ-JJ-IA
    WRITE (LUA-6334) IK-JK-IJ-JK-IN
WRITE (LUA-6334) IL-JL-IJ-JM-IN
 447 CONTINUE
    IF (PHI.EQ. 0.0) GO TO 449
6447 FORMAT (1415)
    WRITE (LUA.6447) IJ.IJ.IJ.JJ.IN.IK.MR.IM.IJ.IJ.II.II.II.IJ
    WRITE (LUA.6447) IK.JJ.IJ.JK.IN.IK.MP.IM.MR.IK.II.II.II.II.
    WRITE (LUA-6447) IL-JK-1J-JM-IN-IK-IL-IK-MP-IL-11-11-11-1J
    GO TO 450
449 CONTINUE
6445 FORMAT (1415)
    WRITE (LUA-6445) IJ-IJ-IJ-JJ-IN-IM-MM-IM-IJ-IJ-II-II-II-IJ
    WRITE (LUA-6445) IK-JJ-IJ-JK-IN-IK-MK-IM-MM-IK-II-II-II-II
    WRITE (LUA-6445) IL-JK-IJ-JM-IN-IK-IL-IM-MK-IL-II-II-II-II-IJ
    69 TO 450
446 CONTINUS
    IF ( 120NE . NE . 3) 50 TO 499
6336 FORMAT (515)
    WRITE (LUA+6336) IJ+IJ+IJ+JK+IN
    WRITE (LUA.6336) IK.JL.IJ.JL.IN
    WRITE (LUA.6336) IL.JM.IJ.JM.IN
451 CONTINUE
    IF(PHI.EG.0.0) 50 TO 453
6448 FORMAT (1415)
    WRITE (LUA.6448) IJ.IJ.IJ.JK.IK.IK.NJ.JI.IJ.IJ.II.II.II.IJ
    WRITE (LUA-644R) IK-IJ-IK-JK-IN-JI-NL-IM-IJ-IJ-II-II-IJ-IJ
    WRITE (LUA-6448) IL-JK-IJ-JL-IK-IK-MG-JI-NJ-IK-II-II-II-II
    WRITE (LUA-6448) IM-JK-1K-JL-IN-JI-MS-IM-NL-IK-II-II-II-II
    WRITE (LUA.6448) IN-JL-IJ-JM-IK-IK-IL-JI-MQ-IL-II-II-II-II
    WRITE (LUA.6448) IPOJLOIKOJNOINOJIOILOIMOMSOILOIIOIIOIIOII
    30 TO 450
453 CONTINUE
6446 FORMAT (1415)
    WRITE (LUA-6446) IJ-IJ-IJ-JK-IN-IK-NN-IM-IJ-IJ-II-II-II-IJ-
WRITE (LUA-6446) IK-JK-IJ-JL-IN-IK-ML-IM-MN-IK-II-II-II-II
    WRITE (LUA-6446) IL-JL-IJ-J4-IN-IK-IL-IM-ML-IL-II-II-II-II
    450 CONTINUE
6444 FOR MAT (F10.4)
    WRITE (LUA.6444) ?
    Z=Z+0Z
    IF (Z.GT.ZMAX) GO TO 999
455 CONTINUS
    1F(120NE.EQ.1) 30 T) 443
 456 CONTINU
    IF (120NE.EQ.2) GO TO 447
457 CONTINUE
    IF ( 120NE . EQ. 3) 50 TO 451
454 CONTINUE
    IF (PHI.LT.THETA) GO TO 441
6333 FORMAT (SA1)
    WRITE (LUA.6333) (IEND(I).1=1.3)
```

STOP STOP STOP

### APPENDIX B

EXAMPLE THREE-DIMENSIONAL FINITE ELEMENT MESH GENERATION INPUT FOR STAR DESIGN

PRECEDING PACE BLANK - NOT FILMED

•	Zone I	Burning	2		
GRAIN MESH					
14 5 39	7 1.00	5			
0.0000 0.000		.1000	-1000	0.0000	25.7143
10.0000 0.000		0.0000	0.0000	0.0000	0.0000
0000		10.0000	10.0000	0.0000	25.7143
0.0000 0.000		0.0000	0.0000	2.0000	0.0000
10-0000 4-815		0.0000	0.0000	0.0000	0.0000
0.0000 0.000		6.5000	5.5000	0.0000	25.7143
2.8548 1.412		0.0000	0.0000	0.0000	0.0000
6.2424 .504	5 6 0	0.0000	0.0000	0.0000	0.0000
6.0000 0.000	10 7 75	•5500	.5500	0.0000	75.0000
0.0000 0.000		0.0000	0.0000	0.0000	0.0000
3.054H 1.432		0.0000	0.0000	0.0000	0.0000_
2.9664 1.428		.0447	.0447	204.7143	256.0000
		0.0000	0.0000	0.0000	0.0000
10.0000 0.000		0.0000	0.0000	0.0000	0.0000
0.0000 0.000		6.1653 0.0000	6.1653	0.0000	25.7143
10.0000 2.282		0.0000	0.0000	0.0000	0.0000
0.0000 0.000		3.2634	3.2634	0.0000	25.7143
3-0816 750		0.0000	0.0000	0.0000	0.0000
6.2653024		0.0000	0.0000	0.0000	0.0000
0.00000.000	0 15 25	2.4726	2.4726	0.0000	25.7143
6.0653 0.000	0 0 0	0.0000	0.0000	0.0000	0.0000
6.6000 0.000	0 16 0	0.0000	0.0000	0.0000	0.0000
0.0000 0.000		5.4398	5.4398	0.0000	25.7143
0.00000.000		0.0000	00000	0.0000_	0.0000
10.0000 0.000		0.0000	0.0000	0.0000	0.0000
6.0424 .497		0.0000	0.0000	0.0000	0.0000
0.0000 0.000		0.0000	0.0000	0.0000	0.0000
10.0000 0.000		0.0000	0.0000	0.0000	0.0000
3.0816739		0.0000	0.0000	0.0000	0.0000
6.5734 .266		0.0000	0.0000	0.0000	0.0000
0.000 0.000		0.0000	0.0000	0.0000	0.0000
3.0462 1.435		0.0000	0.0000	0.0000	0.0000
2.8462 1.377	4 0 0	0.0000	0.0000	0.0000	0.0000
6 • 1653 1 • 751		0.0000	0.0000	0.0000	0.0000
5.8194 1.790		0.0000	0.0000	0.0000	0.0000
6.4370 1.380		0.0000	0.0000	0.0000	0.0000
2.5000 .570		0.0000	0.0000	0.0000	0.0000
10.0000 2.282		0.0000	0.0000	0.0000	0.0000
10.0000 0.000		0.0000	0.0000	0.0000	0.0000
		0.0000	0.0000	0.0000	0.0000
10.0000 2.282		0.0000	0.0000	0.0000	0.0000
9.0000 0.000		0.0000	0.0000	0.0000	0.0000
10.0000 3.499		0.0000	0.0000	0.0000	0.0000
2.8462 1.389		0.0000	0.0000	0.0000	0.0000
6.1107 1.371	4 40 0	0.0000	0.0000	0.0000	0.0000
6.5500 0.000	0 0	0.0000	0.0000	0.0000	0.0000
6.4985 .232		0.0000	0.0000	0.0000	0.0000
6,3535 .421		0.0000	0.0000	0.0000	0.0000
6.1424 .531		0.0000	0.0000	0.0000	0.0000
4.5235 .965		0.0000	0.0000	0.0000	0.0000
2.9046 1.398		0.0000	0.0000	0.0000	0.0000
6.0000 0.000		-2-2899	-2 -2899	0.0000	75.0000_
5.8222 0.000		0.0000	0.0000	0.0000	0.0000
5.6476 .374		0.0000	0.0000	0.0000	0.0000
5.4599472		0.0000	0.0000	0.0000	0.0000
4.0209 .851	78 0 0	0.0000	0.0000	0.0000	0.0000
2.5819 1.243		0.0000	0.0000	0.0000	0.0000

6.0000	0.0000	45	75	-2 -1	0354	-2 .0	354	0.00	00	75.0000
6.5500	0.0000	0	0	0.1	0000	0.00	000	0.00	0.0	0.0000
6.5500	0.0000	0	0		0000	0.00	200	0.00		0.0000
6.4985	.2324	Ö	Ö		0000	0.00		0.00		0.0000
						0.00			TR. 1977	
6.3535	.4213	0	0		0000			0.00		0.0000
6a1424	.5313	0	0		0000	0.00		0.00		0.0000
4.5235	.9650	0	0		0000	0.00	303	0.00	(B)(B)	0.0000
2.9046	1.3988	46	0	0.0	0000	0.00	000	0.00	00	0.0000
0.0000	0.0000	47	0	3.1	7101	3.71	01	0.00	00	0.0000
5.8222	0.0000	0	0	2000	0000	0.00	00 50 70 -	0.00	F197732	0.0000
5.8222	0.0000	o	0		0000	0.00		0.00		0.0000
						0.00		0.00		0.0000
5-7764	-5066	0	0		0000				-	
5.6476	.3745	0	0		0000	0.00		0.00		0.0000
5.4599	.4722	0	0		0000	0.00		0.00	50.000	0.0000
4.0209	.8578	0	0	0.0	0000	0.00	00	0.00	00	0.0000
2.5819	1.2434	49	0	0.0	0000	0.00	00	0.00	00	0.0000
6.0000	0.0000	49	75	-2.2	2899	-2.25	99	0.00	00	75.0000
0_0000	0.0000	51	0		979	3.29	79	0.00	0.0	0.0000
6.0000	0.0000	53	75	-2.0		-2.0		0.00		75.0000
	1 4	5	1772		333	-2.00.		0.00		
1 1 5		2								BUSCE
	The second second second									
2 5	3 5	5								
2 6	3 9	3								
210	1 10	3								
3 5	4 14	5			-1,000					
	1 14	3								
1 1	1 5	2	2	13	12	1	1	0	0	0 0
2 i	2 5	3	12	13		i	i	0	0	0 0
3 1	3 4	4		15		i	i	Ö	0	0 0
	100		8		26				1357	
	4 4	5	26	15	. •	1	1	. D	0 -	-0 -0-
5 5	1 9	2	2	17	24	13	1	0	0	0 0
. b	3 5	4	8	9	26	15	2	0	0	0 _ 0 _
7 •	4 5	5	26	9		15	2	0	0	0 C
. 8 . 5	2 9	3	24	17	6	13	2	0	0	0 0
9 9	1 10	2	2	7	24	17	2	0	0	0 0
	2 10	. 3	24	7	6	17	2	.0	0 .	0 0
	3 9		6	17	40	9	3	0	0	0 0
		5						7	207	
12 5			40	17		9	3	0	0	
13 9	3 10	4	6	11	40	17	3	0	0	0 0
. 14 . 9	4 10	5	40	11	4	17	3	0	0	0 . 0
15 10	1 14	2	5	3	55	7	3	0	0	0 0
16 . 10 .	2 14	3	22	3	20	7	3	. D.	D	1
17 10	3 14		20	3	32	11	3	0 .	0	0 0
. 18 10	4 14	5	32	3	4	11	3	0	0	0 D
-1 0	0 0	0	0	0	0	0	0	0	0	0 0
0.0000	0000		10000		A FIFTH					CONTRACTOR .
1 1	1 5	2	2	13	12	1	1	0	0	0 0
		3						0		
	2 5		12	13	8	1 -	1		0	
3 1	3 4		8	15	26	1	1	0	0	0 0
4 1	4 4	5	26	15	4	1	1	0	0	0 . 0
. 5	1 9	?	5	17	24	13	1	0	0	0 0
5 4	3 5	4	8	9	26	15	2	0	0	0 0
7 4	4 5	5	26	9		15	2	0	0	0 0
A .5	2 9		24	17		13	2	0	0	0 0
9 9 10 9 11 5 12 5 13 9	2 9 1 10 2 10 3 9 4 9 3 10 4 10 1 14 2 14 3 14	3	24 2 24 6 40 6 40 2 22	17 7 7 17 17	6 24 6 40 4 22	13 17 17 17 9 9	2	0	000000000000000000000000000000000000000	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
	1 10				• ;		-			
9 9 10 9 11 5 12 5 13 9 14 9 15 10 16 10	2 10	3				.,				
11 5	3 9	•		17	40	9	3	U	U	0 0
12 5	4 9	5	40	17	4	9	3	0	0	0 0
13 9	3 10	4	6	11	40	17	3	0	0	0 0
14 . 9.	4 10	. 5	40	11	4	17	3	0	0	C 0
15 10	1 14	2	2	3	22	7	3	. 0	0	0 0
15 10 16 10	2 14	3	22	3	32	7 7 11	3	0	0	0 0
17 10	1 14		20	•	32	11	1	0	0	0 0
14 10	4 11		33		32	ii	1			0 0
17 10 18 10 -1 0	3 14 4 14 0 0	323454523450	20 32 0	11 11 3 3 3 3	0	0	2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
-1 0	0 0	U	U	C	U	0	U	U	U	0 0
2.0000										

1 1	1 5	2	2	13	12	1	1	0	0	٥	. (	,
1 1 2 1 3 1 4 1 5 5 5		3	12	13	,		i	0	ō	Ö		
3 1			A 26	15	26	1	1	0	0		(	)
3 1 4 1 5 5 6 4 7 4 8 5 9 9 10 9 11 5 12 5 13 9 14 9 15 10 16 10 17 10	1 9 3 5	5	26	15	24	13	1	0	0	0	(	)
	1 9	5	2	17	54	13	1	0	0	0	(	)
W. Commercial St. Victoria	3 5	•	56	9	26	15	5	0	0	0		
		4 5 3 2 3 4 5 4 5 2 3 4 5 0	26	17	6	15	2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	0	0	000000000000000000000000000000000000000	000000000000000000000000000000000000000	
	2 9 1 10 2 10 3 9		24	1,	24	13	•		0	0		Tar.
10 9	2 10	-	24	7 7 17 11 11 11 3 3 5 0	6	17 17 9 9 17	2	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0	0		
11 5	3 9	4	6	17	40	9	3	ō	0	ō	ò	
125	3 10	5	6	17	•	9	3	0	. 0	.0		_
13 9	4 9 3 10 4 10 1 14	4	6	11	*0	17	3	0	0 0	0	0	
_ 19 9	• 10	5	•0	11		17	3	0	0	0	0	
15 10	1 14	2	5 5		5.5	?		•	0	0	C	
16. 10	2 14 3 14	3	50	3	50	'	3	0	0	0	0	
17 10			32		35	11		0	0	0	0	
-1 0	1 1 4 2 1 4 5 1 4 6 0 0	0	0	0	32	7 7 11 11 0	0	0	0	0		
-1 0						•		·			·	
1 1	1 5	2	2	13	12	1	1	0	0	0	0	
2 1	2 5	3	12	13	8 26	1	1	0	0	0	. 0	
3 1	3 •	•	8	15	26	1	1	0	0	0	0	
		5	26	13 13 15 15	24	1	1 -	0 .	0	. 0	0	
5 5	1 9	5	2	17	54	13	1	0	0	0	9	
	3 5	•	26		26	15	2	0	0	.0.	-	
9 3	2 9	•	24	17		13	2	0	0 0	0		
1 1 2 1 3 1 5 5 6 4 7 4 8 5 9 9	1 10	2	26 24 2 24	7	26 6 24	17	2	0 0 0	o	ŏ	·	
10 9	2 _ 10 _		24	7	. 6	17	2	. 0	0	0		
11 5 12 5 13 9 14 9 15 10	1 5 2 5 3 4 4 1 9 3 5 4 5 2 9 1 10 2 10 3 9	2545245222454525450	6	9 17 7 7 17	40	1 13 15 15 13 17 17 9 9 17 17 7	1 1 1 2 2 2 2 2 3 3 3 3 3 3 3 3 3 3 3 3	0	0	0	0	
12 5	4 9	5	40	17	•	9	3	0	0	0	. 0	
13 9	3 10	•	6	11	40	17		0	0	0	0	1
14 9	1 14	5	40	11	22	17		0	0	0	9	-
15 10	1 14 2 19		6 40 2 22		20		3	0	0	0		
17 10	1 14		50	3	32	11	3	Ü	0	0	-	
18 10	. 10	5	32	3 3 3 0	•	ii	3	0	0	0	1	
-1 0	0 0	0	0	0	0	0	0	0	0	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		
18 10 -1 0 6-0000												
1 1	1 5 2 5 3 4 4 4 1 9 5 5 5 5 4 5 2 9 1 10 2 10	234524532345	5	13	12	1	1	0	0	0		,
3 1	2 . 5 .	3	12	13	8	1	1	0	0	0		Contract Contract
3 1	:		8	15	56	1	1	0	0	0		
	2 5 3 4 4 4 1 9 3 5 4 5	2	20	13 15 15 17	24	13		0 0 0	0	0		
	3 5		2 26 24 2	9	26	15	2	0	0	0		
7 4	4 5	5	26	9 17 7 17 17	•	15 15 13 17	2	0	o	Ö	ò	
A 5	2 9 1 10 2 10	3	24	17	6 6 6	13	2	0	0		0	
9 9	1 10	2	2	7	24	17	5	0	0	0	0	
10 9	2 10	3	24	7	6	17	5		0	0	0	
11 5	3 9	•	6	17	40	9	3	0	0	0	0	
15 5	3 9 3 10	3	6	17	+0	9 17 17	1 1 2 2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3	0	0	000000000000000000000000000000000000000	0	
13 9	10		40	11	*0	17		0	0	0	,	
15 10	1 10	5 2	2	**	22	,	;	0 .	0			Miller Will
11 5 12 5 13 9 14 9 15 10 16 10 17 10 18 10 -1 0 6.0000 1 1 2 1 3 1 4 1 5 5 6 4 7 4 8 5 9 9 10 9 11 5 12 5 13 9 15 10 16 10 17 10 16 10	2 14	3	22	1	20	7 7 11	3	Ö	0		,	
17 10	3 14		50	3	32	11	•	0	0	0	i	,
18 10		50	32	3	•	11	3	0	0	0		
-1 0	0 0	0	0	0	0	0	0	0	0	0	(	
00000			-									

# Zone 2 Burning

		Polic				
GRAIN MESH						
19 5	39 3	1.0				
0.0000	0.0000	1 2			0.0000	25.7143
0.0000	0.0000		0.000		0.0000	0.0000
10.0000	0.0000		0.000		0.0000	0.0000
0.0000	0.0000	3 2			0.0000	25.7143
0.0000	0.0000		0.000		0.0000	0.0000
10.0000	4.8157		0.000		0.0000	0.0000
0.0000	0.0000	5 2			0.0000	25.7143
3-2301	2.8127		0 0.000		0.0000	0.0000
6.6176	1.9051		0.000		0.0000	0.0000
6.0000	0.0000	7 7	7 2 2 2 2		0.0000	75.0000
●.0000	0.0000		0 0.000		0.0000	0.0000
2.9664	2.8696	8	0.000		204.7143	256.0000
- 0.0000	1.4285		0 0.000		0.0000	20000
10.0000	0.0000		0 0.000		0.0000	0.0000
0.0000	0.0000	11 2			0.0000	25.7143
0.0000	0.0000		0 0.000		0.0000	0.0000
10.0000	2.2824		0 0.000		0.0000	0.0000
0.0000	0.0000	13 2			0.0000	25.7143
	1-0038		0.000		0.0000	- 0.0000
6.8979	0377		0 0.000		0.0000	0.0000
0.0000	0.0000	15 2			0.0000	25.7143
6.6979	0.0000		0 0.000		0.0000	0.0000
6.6000	0.0000		0 0.000		0.0000	0.0000
0.0000	0.0000	17 2			0.0000	25.7143
	_ 0.0000		0 0.000		0.0000	0.0000
10.0000	0.0000		0 0.000		0.0000	0.0000
6.4176	1.8235		0 0.000		0.0000	0.0000
10.0000	2.9640		0 0.000		0.0000	0.0000
0.0000	0.0000		0 0.000		0.0000	0.0000
10.0000	1.4820		0 0.000		0.0000	0.0000
401329	9612		0.000		0.0000	0.0000
7.7478	1.1383		0 0.000	0 0.0000	0.0000	0.0000
0.0000	0.0000	0	0 0.000	0.0000	0.0000	0.0000
3.6997	2.7576	26	0 0.000	0.0000	0.0000	0.0000
3.4997	2.6718	C	0 0.000	0 .0000	0.0000	0.0000
6.7979	3.0434	28	0.000	0 .0000	0.0000	0.0000
6.0154	3.6558	0	0.000	0 0.0000	0.0000	0.0000
6.4370	.7572	30	0 0.000	0 0.0000	0.0000	0.0000
2.5000	.5706		0 0.000	0.0000	0.0000	0.0000
10.0000	2.2824		0.000	0.0000	0.0000	0.0000
0.0000	0.0000		0 .0.000		0.0000	0.0000
10.0000	1.1267		0.000		0.0000	0.0000
0.0000	0.0000		0.000		0.0000	0.0000
10.0000	2.2824		0.000		0.0000	0.0000
0.0000	0.0000		0.000		0.0000	0.0000
10.0000	3.4992		0.000		0.0000	0.0000
3.4997	2.7217		0 0.000		0.0000	0.0000
6.7275	1.4740		0.000		0.0000	0.0000
- 8.0000	0.0000		0.000		0.0000	0.0000
7.8126	.8452	0	0 0.000		0.0000	0.0000
7.2856	1.5321		0.000		0.0000	0.0000
6.5176	1.9319		0.000		0.0000	0.0000
5.7125	2.1476		0.000		0.0000	0.0000
4.9074	2.3633		0.000		0.0000	0.0000
- 6.0000	0.0000	. 45 7			0.0000	75.0000
7-1111	0.0000		0.000		0.0000	0.0000
6.9445	.7513				0.0000	0.0000
6.4761	1.3619		0 0.000		0.0000	0.0000
5.7935	1.7172			100 A 100 COMPANY	0.0000	0.0000
5.0778					0.0000	0.0000
4.3622	2.1007	••	0 0.000	0.0000	0.0000	3.0000

6.0000	0.0	000	45	75	7	466	1	466	0.0	000	75.0	0.00
8.0000	0.0	000	0	0	0.0	000	0.0	2000	0.0	000	0.0	000
8.0000		000	0	0		000	180 200	000		000		000
7.8126		452	0	0		000		000		000		000
7.2856		321	0	C		300		000		000		000
6.5176		319	0	Ö	_	000		000		000		000
5.7125		476	õ	Ö		000		000		000		000
4.9074		633	46	0		000		0000		000	0.00	000
	7000											
0.0000	200	000	47	0		601		601		000		000
7.1111		000	3	0		000		000		000	GOTAL P.	000
7.1111		000	0	0		000		000		000		000
6.9445		513	0	0		000		000		000	. 0.0	
6.4761		619	0	0		000		000		000		000
5.7935	1.7	172	0	0	0.0	COO	0.0	000	0.0	000	0.0	000
5.0778	1.9	090	0	0	0.0	000	0.0	000	0.0	000	0.0	000
4.3522	2.1	007	48	0	0.0	000	0.0	000	0.0	000	0.0	000
6.0000	0.0	000	49	75	A	399	8	399	0.0	000	75.0	000
	0.0	000	. 51	0	4.5	968	4.5	868	0.0	000	-0.0	000
6.0000	0.0	000	53	75	7	466		466	0.0	000	75.0	000 -
1 1	1	11	5									
2 12	ī	12	5									By Barrier
3 13	i	14	5									
1 1	i	11	5	2	44		1	1	0	0	0	0
211	i.	12	5	2	42		44	2	a	0		_ a
3 12	1	14	5	2	3		42	3	0	0	0	0
-1 0	0	0	ć	5	0	0	0	0	0	0	0	
0.0000					u	u	U	U				
					44		1				0	
1 1	1	11	5	2 2		4		1	0	0	0	0
2 11	1	12	5		42	•	44	2	0			
	1 .	14	5	2	3	•	42	3	0 .	. 0		0
-1 0	C	0	C	0	0	0	0	C	0	0	0	0
2.0000											2.0	
1 1	1	11	- 5	5	44	•	1	1	0	0	0	0
2 11	1	12	5	2	42	•	44	2	0	0	Q	0
3 12	1	14	5	2	3	•	42	3	0	0	0	0
-10	0	0	0	C	0	0	0	0	0	0	.0.	
4.0000												
1 1	1	11	5	2	44	•	1	1	0	0	0	C
2 11	1	12	5	2	42		44	2	0	0	0	0
3 12	1	14	5	2	3		42	3	0	0	0	0
-1 0	0	0	0	0	0	0	0	0	0	0	0	0
6.0000	Will be	4.		1004			1.7.8			1 48		110
1 1	1	11	5	2	44		1	1	0	0	0	0
2 11	i	12	5	2	42		44	2	0	0	0	0
3 12	i	14	5	2	3.		42	3	Ö	Ö	Č	0
-1 0	Ô	0	ŏ	0	0	0	0	0	0	Ö	ő	Ö
8.0000		·				0			U			
0.0000												

# **Zone 3 Burning**

		Tolle 3	Dui II	III		
GRAIN MESH						
19 5	30 3	1.00	•			
0.0000	0.0000	1 25	.1000	-1000	0.0000	25.7143
0.0000	0.0000	0 0	200000	0.0000	0.0000	0.0000
10.0000	0.0000	5 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	3 25	10.0000	10.0000	0.0000	25.7143
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	4.8157	• 0	0.6000	0.0000	0-0000	0.0000
0.0000	0.0000	5 25	0.0000	6.5000	0.0000	25.7143
7.0059	3.3539	5 0	0.0000	0.0000	0.0000	0.0000
-6.0000	0.0000	7 75	3.5000	3.5000	0.0000	75.0000
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
3.8163	4.3487	8 0	0.0000	0.0000	0.0000	0.0000
2.9664	1.4285	9 49	-2.9053	-2.9053	204.7143	256.0000
	_0.0000	0. 0	0.0000	0.0000	0.0000	0.0000
10.0000	0.0000	10 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	11 25	7.6890	7.6890	0.0000	25.7143
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	2.2824	12 0	5.6356	0.0000	0.0000	0.0000
0.0000	0.0000	13 25	0.0000	0.6356	0.0000	25.7143
7.7890	0571	14 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	15 25	4.2517	4.2517	0.0000	25.7143
7.5890	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
6.6000	0.0000	16 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	17 25	7.1756	7.1756	0.0000	25.7143
0.0000	- 0.0000	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	0.0000	18 0	0.0000	0.0000	0.0000	0.0000
6.8059	3.2120	0 0	0.0000	0.0000	0.0000	. 0.0000
10.0000	0.0000	20 0	0.0000	0.0000	0.0000	0.0000
10.0000	2.4477	22 0	0.0000	0.0000	0.0000	0.0000
	1.2268	. 0 0	0.0000	0.0000	9.0000	0.000
8.8653	2.1728	24 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
4.3757	4.1162	24 0	0.0000	0.0000	0.0000	0.0000
4.1757	4.0106	0 0	0.0000	0.0000	0.0000	0.0000
7.6890	4.4067	28 0	0.0000	0.0000	0.0000	0.0000
6.4370	2069	30 0	0.0000	0.0000	0.0000	0.0000
2.5000	5.9147 .5706	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	2.2824	32 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	1.1267	34 0	0.0000	0.0000	0.0000	0.0000
	0.000.0	0 0	0.0000	0.0000	0.0000	0.0000
10.0000	2.2824	36 0	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	0 0	0.0000	0.0000	0.0000	2000.0
10.0000	3.4992	38 0	0.0000	0.0000	0.0000	0.0000
7.5962	4.0939	0 0	0.0000	0.0000	0.0000	0.0000
945000	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
9.1721	1.4792	0 0	0.0000	0.0000	0.0000	0.0000
8.2498	2.6812	0 0	0.0000	0.0000	0.0000	0.0000
6.9059	3.3807	0 0	0.0000	0.0000	0.0000	0.0000
6.9426	3.3709	0 0	0.0000	0.0000	0.0000	0.0000
6.9793	3.3611	45 0	0.0000	0.0000	0.0000	0.0000
6.0000	0.0000	43 75	.6601	.6601	0.0000	75.0000
8.4444	0.0000	0 0	0.0000	0.0000	0.0000	0.0000
8.1530	1.3148	0 0	0.0000	0.0000	0.0000	0.0000
7.3331	3.0051	0 0	0.0000	0.0000	0.0000	0.0000
6.1712	2.9964	0 3	0.0000	0.0000	0.0000	0.0000
6.2038	2.9876	44 0	0.0000	0.0000	0.0000	0.0000
						The second secon

6.0000	0.0000	45	75	.5	968	.5	860	0.0	000	75.0	000
9.5000	0.0000	0	0		000		000		000		000
9.5000	0.0000	Ö	0	13.00	000		000		000		000
9.1721	1.4792	0	0		000		000	-	000		000
8.2498	2.6812	0	Č	1.0	000		303		000		000
6.9059	3.3807	5	0		000		000		000		200
6.9425	3.3709	0					000		000		000
		-	0		000						
6.9793	3.3611	46	0	1.00	000	2000	000		000		000
0.0000	0.0000	47	0		601	6.6	-	0.0			000
8.4444	0.0000	0	0		000	0.0			000		000
8.4444	0.0000	0	0	0.0	000	0.0	000		000	0.0	000
8-1530	1.3148	. 0	0	0.0	000	0.0	000	0.0	000	0.0	000
7.3331	2.3832	0	0	0.0	000	0.0	000	0.0	000	0.0	000
6.1385	3.0051	0	0	0.0	000	0.0	000	0.0	000	0.0	000
6.1712	2.9964	0	0	0.0	000	0.0	000	0.0	000	0.0	000
6.2038	2.9976	48	0	0.0	000	0.0	000	0.0	000	0.0	000
6.0000	0.0000	49	75		601	10.00	601	0.0		75.0	
0.0000	-0-0000	51	. 0	5.9		5.9		0.0		0.0	
6.0000	0.0000	53	75		869		868	0.0		75.0	
1 1	1 12	5	' '	• •	001	• •	000	0.0			
2 13											
	1 13	5									
314	1 14	5									2 .
1 1	1 12	5	2	45	4	1	1	0	0	0	0
2 -12	1 . 13.		2	43	. 4	45	2	0.	0 -	- 0	
3 13	1 14	5	5	. 3	4	43	3	0	0	0	0
1. 0	0 0	0	0	0	0	0	. 0	0	0 .	0	0
0.0000											
_ 1 _ 1	1 12	5	2	45	4	1	1	0	0	0.	0
2 12	1 13	5	2	43		45	2	0	0	0	0
313	1 . 14	5	2	3	4	43	3 .	0	_ Q	. 0	
-1 0	0 0	0	0	0	0	0	0	0	0	0	C
2.0000											
1 1	1 12	5	2	45	4	1	1	0	0	0	0
12	1 13	5	2	43	•	45	2	o	0	Ö	o o
3 13	1 14	5	2	3	4	43	3	0	Ö	Ö	0
-1_0_	0 0	ó	0	0	0	0	0	ő	.0.	0	_ G
4.0000	u u	. 0	U	U	U	0	U	U			L
1. 1	1 12	5	2	45	4	1	1	0	0	0	0
2 12	1 13	5	2	43	4	45	2	0	0	0	0
3 13	1 14	5	2	3	4	43	3	0	0	0	. 0
-1 0	0 0	0	0	0	0	0	0	0	0	0	0
6.0000											
1 1	1 12	5	5	45		1	1	0	0	0	0
2 12	1 13	5	2	43	4	45	2	0	0	0	.0
3 13	1 14	5	2	3	4	43	3	0	0	0	0
-1 0	0 0	0	0	C.	C	0	0	0	0	0	0
8.0000											

# DISTRIBUTION

	No. of
	No. of Copies
Defense Documentation Center	
Cameron Station	
Alexandria, Virginia 22314	12
IIT Research Institute	
ATTN: GACIAC	
10 West 35th Street	1 AFTR DOF BRIEfon
Chicago, Illinois 60616	
US Army Materiel Systems Analysis Activity	
ATTN: DRXSY-MP	2 THE GOLD W. Process
Aberdeen Proving Ground, Maryland 21005	
CPIA Distribution	96
Vanderbilt University	
Department of Civil Engineering and Engineeri	ng Science
ATTN: Dr. R. M. Hackett	5
Box 1537 Station B	TOWN area 7 noon continued to
Nashville, Tennessee 37235	
Air Force Rocket Propulsion Laboratory	NASA Alamell Space Right Center
ATTN: DYSC/Mr. W. Andrepont	brolle-kontil 8 na 1977.
ÇAPT J. Donn	ethype3 Data 1
Mr. J. Levine	
Edwards, California 93523	1 more all a second
Hercules, Inc.	
ATTN: Mr. N. Peterson	ANNOS J. B. M. L. P. L.
P.O. Box 98	DREE MARTINESS RATIO graduated
Magna, Utah 84044	2.0.364 map
Chemical Systems Division	
Inited Technologies	
ATTN: Dr. R. S. Brown	Department of Machenical Engineering
Mr. R. Waugh	panasil A. G. a. defor
Sunnyvale, California 94088	Mail Ealer City, 1995 BY 152

## **DISTRIBUTION**

	No. of Copies
California Institute of Technology ATTN: Dr. F. E. C. Culick 204 Karman Laboratory/Mail No. 301-46 1201 East California Pasadena, California 91109	ntaig 1
Naval Weapons Center Aerothermochemistry Division ATTN: Code 608, Dr. R. L. Derr Mr. C. J. Bicker China Lake, California 93555	Ceronse Documentón Center Comerca Station Alexatoria, Virgina 2,314
California State University Department of Mechanical Engineering ATTN: Dr. F. Reardon Sacramento, California 95819	Land Burder The Carlos of the
Georgia Institute of Technology Department of Aerospace Engineering ATTN: Dr. E. W. Price Dr. B. T. Zinn Atlanta, Georgia 30332	LUS Amily Material Element Analysis Author ALTIN DELYSYAND Abordeen Proving Strandi, Maryteini 21001
Rockwell International Corporation Rocketdyne Division ATTN: Mr. W. T. Brooks P.O. Box 548 McGregor, Texas 76657	Vanderell' University Cop. I gent of Cos.I Engineering and English ACTA Co. T. M. Hackelt Box 1637 Scripto B
NASA-Marshall Space Flight Center ATTN: Mr. B. Shackelford Mr. C. Forsythe EP25 Huntsville, Alabama 35812	All Forces Rucket Proposition Laboratory  LTM: DYSC Mr. v. andrepont  CAPT II Strop  Laboratory  Cameral California 84523
Aerojet Solid Propulsion Company ATTN: Mr. R. L. Lovine Building 2019/Department 4350 P.O. Box 13400 Sacramento, California 95813	Hercelas, Inc. 1771 Mt. N. Peterson 2.0 Box 98 Sagna, Utser 84641
University of Utah Department of Mechanical Engineering ATTN: Dr. G. A. Flandro Salt Lake City, Utah 84112	neva 2 emetavá ledimedů velpodroba i edmougles ATTA CA P. STENAN VEL B. Woudh

## DISTRIBUTION

	No. of Copies
Thiokol Chemical Corporation	
Wasatch Division	
ATTN: Dr. C. M. Milfeith	1
Mr. S. Folkman	1
P.O. Box 524	
Brigham City, Utah 94302	
Thiokol Chemical Corporation	
Huntsville Division	
ATTN: Mr. G. Estes	1
Mr. K. Herren	1
Huntsville, Alabama 35807	
Princeton University	
Department of Aerospace and Mechanical Science	
ATTN: Dr. W. A. Sirignano	1
P.O. Box 710	
Princeton, New Jersey 08540	
Commander	
US Army Materiel and Readiness Command	
ATTN: DRCRD	1
DRCDL	1
5001 Eisenhower Avenue	
Alexandria, Virginia 22333	
Battelle Columbus Laboratories	•
Durham Office	1
P.O. Box 8796	
Durham, North Carolina 27707	
Hercules, Inc.	
Allegany Ballistics Laboratory	
ATTN: Mr. J. Murray	1
P.O. Box 210	
Cumberland, Maryland 21502	
DRSMI-FR, Mr. Strickland	1
-LP. Mr. Voigt	1

## **DISTRIBUTION** (Concluded)

	No. o Copies
DRDMI-X	1 companies demonstrated
-T, Dr. Kobler	1
-TKP	STATUTE OF THE PARTY OF
-TKC	1
-TKK	1
-TKA	10
-TBD	3
-TI (Record Set) (Reference Copy)	Haur Commission Room